ASK 13

Flight- and Maintenance Manual

ALEXANDER SCHLEICHER SEGELFLUGZEUGBAU

POPPENHAUSEN /RHÖN

Flight - and Maintenance - Manual

for the Glider

AS - K 13

Edition

This Handbook has to be carried on bord of the aircraft.

It belongs to the glider

AS - K 13

.....

Serial No. 13383.....

Manufacturer Alexander SCHLEICHER..

.....

Holder PPBS, Aérodrome de la foret.....

71600 PARAY LE MONIAL.....

.....

Record of Revisions

Rev No.	Pages Affected	Date of Issue	Date of Approval	Date of Insertion	Ref. / Signature
TN 4	22	04.07.69	05.08.69	20.08.69	Juw
TN 7	11a	01.11.75	03.12.75	12.12.75	Juw
TN 7a	4b	21.01.87	17.02.87	26.02.87	Juw
TN 12	annex (elevator)	04.10.89	17.10.89	25.10.89	Juw
TN 14	annex (dive brake assembly)	27.09.91	07.10.91	16.10.91	Juw
TN 15	9a	20.12.05	02.03.06	16.06.06	mg
TN 18	annex (elevator)	25.05.10	17.05.10	27.05.10	mg

1. Operating Limitations

Air speeds:

Max. speed	125	mph
rough air	87	mph
aero tow	87	mph
auto and winch tow	62	mph

Weights:

Empty weight	650	lbs
Max. weight	1060	lbs
Max. weight of non lifting parts	710	lbs

Kategory: 2 BVS

Limit load factor

up	4,0
down	−2,0

Safety factor 2,0

Center of gravity position in flight:

Leveling means	Tangente to rib Nr. 3 horizontal.
Datum	wing leading edge rib 3
Max. forward	2,75 behind datum point
Max. rearward	9,7 " " "

Weak link in the tow cable:

Winch tow	max.	235o lbs
	min.	185o lbs
Aero tow	max.	158o lbs
	min.	1o6o lbs

2. Operating Directions:

Winch tow:

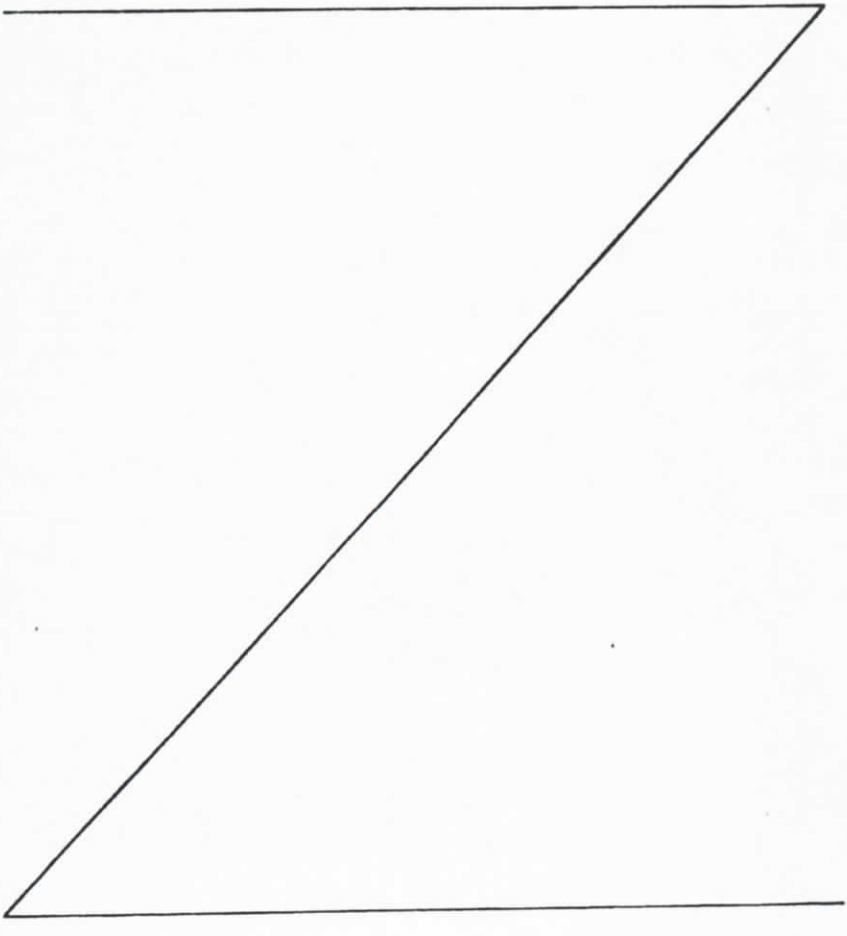
Max. tow speed is 62 mph.

Attention. In winch tow pulling the stick back means increase of speed. When lifting off ease the stick some what to overcome a light tendency to pitch up. Best attitude in climb is with stick normal. Winch tow on the belly hook only.

-ASK 13- Flughandbuch

Gewichtstrimmung

Das Befestigen eines 3,8 kp schweren Trimmgewichtes,mit zwei Rändelschrauben,ergibt einen Ausgleich für 5 kp Führergewicht im vorderen Sitz.
Es können max. 2 Trimmgewichte Befestigt werden.



Aero tow

Max. towing speed is 87 mph.
For aero tow the nose hook is preferable.
Pull release till the stop.

Before every take off check canopy and airbrakes for complete locking.

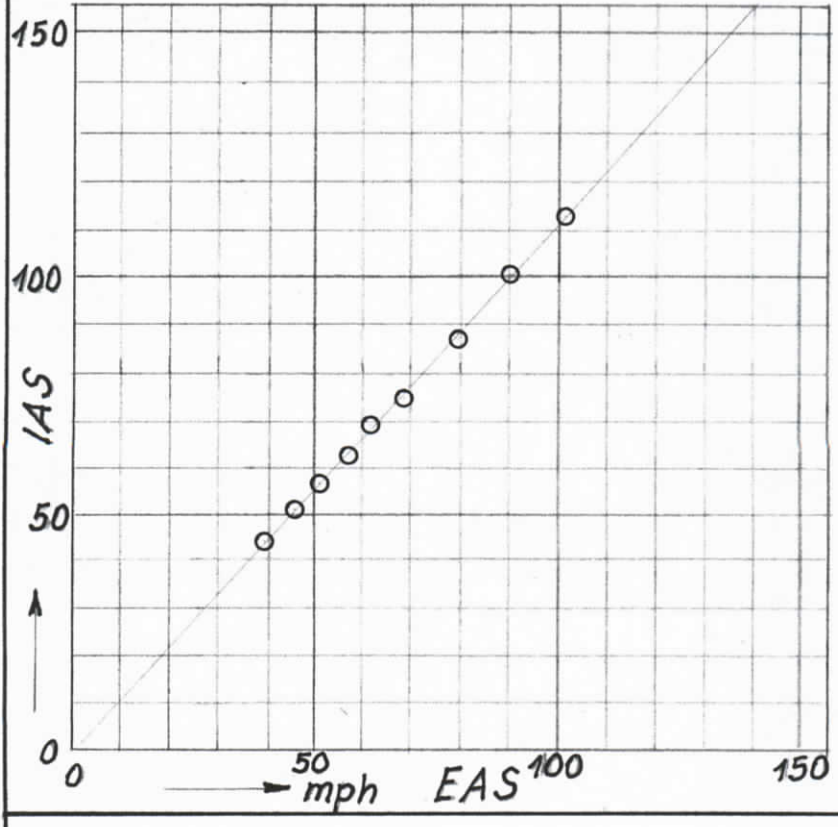
Adjustment of the front rudder pedals.

Pull back the pedals with the heels and lock the adjusting link to the desired position.
Adjusting is possible during flight too.

In flight:

The presented values are calculated. They are related to EAS. There has to be considered the position error, see Fig. 1.
The indicated air speed reading may drop to zero when the glider is slipping or skidding due to movement of the total pressure peak from the nose to the side.

Fig. 1



Stalling speed with an all up weight

of	84o	lbs	is	35 mph
	1o4o	lbs	is	38 mph

Speed at	min	sink	4o mph
	optim.	glide	5o mph

Landing:

Approach speed 5o - 55 mph.
The glide angle can be adjusted in a wide range with the airbrakes. Touch down is best with partly extended airbrakes only. The wheel brake is actuated by the airbrake lever when fully pulled back.

Stalling and Spinning Behaviour:

With stick full back the aircraft can be controlled by the rudder. Applying a large amount of rudder will cause a spin. There has to be considered the influence of the center of gravity position to the spinning characteristics. With the C.of Gr. pos. forward the aircraft will tend to go into a spiral dive and build speed very rapidly. In this case the airbrakes have to be opened first before pulling out.

With C. of Gr well in the middle spinning is normal and the aircraft will recover by giving free the controlls allone.

With C.of. Gr. near the most rear ward position spin recovery has to be managed by the following standard methode:

a) apply opposite rudder (i.e. against the direction of rotation of the spin);

b) pause ;

c) ease the control column forward until the rotation ceases

d) centralise rudder and allow aircraft to dive out.

At high speed there has to be watched the speed limits. When a speed of 87 mph is surpassed involuntary the airbrakes should be opened slowly.

Remember: At higher speeds the airbrake lever force is actuating in opening sense.

Rain drops, hoar frost and ice will disturb the wing surface, so quite adverse flying characteristics may result. There fore caution is advised in such cases during approach, give enough speed margin.

Emergency Jettisoning of Canopy: To bail out the canopy has to be opened at the release knob at the left side and pushed forward out of the hinge. The hinges have to be watched for easy moveability.*

Cloud flying

The glider has sufficient strength for cloud flying. Nevertheless some principal rules should be considered:

* TN-no. 5 from 17.12.70:

"To bail out the canopy has to be opened by pulling both release knobs on left and right side."

1) Overspeed in cloud flying should be prevented in any case. There should be the rule to open the airbrakes early at speeds of 65 to 75 mph.

2) Minimum equipment for cloud flying:
Air speed indicator with pitot tube protected against icing.
Sensitiv altimeter
Variometer
Compass
Turn and bank (power source insensitive against icing).
Chronometer
An artificiel horizon and accelerometer is recommended.

3) The ATC rules are to be observed.

4) Minimum Equipment:

Airspeed indicator ranging from 3o to 125 mph.

Altimeter
Safety belt and shoulder harness.
Back cushion if no parachute is carried (min. thickness 4 inch).
Balance - and data plate
Flight Manual.

5) Adjusting Data:

The adjusting and washout - angles as well as the control surface deflections are shown in the outline drawing.
At repairs care should be taken to observe the tolerances.
By the particular kinemetics of the control mechanism the aileron deflection will be influenced by the elevator. With normal stick position the ailerons have to be normal.

Flugbetrieb mit offener Haube

- Bei der Inbetriebnahme des Segelflugzeuges mit offener Haube, sind die Abschottplatten am Holm- bzw. Sitzspant anzubringen.
- Die geschlossene Haube ist gegen die offene Haube auszutauschen.
- Der Haubennotabwurf der offenen Haube erfolgt wie bei der geschlossenen Haube (weiße und rote Knöpfe).
- **Schutzbrille tragen**!
- **Beachte: höhere Sinkwerte!**

With pushed and pulled stick the ailerons are some what zoomed.

The controls have stops:

Rudder control: Fixed stop at the lower hinge.

Aileron control: Fixed stop at two hard wood pieces down the front seat.

Elevator control: Backward - fixed stop at the front edge of the seat, forward - fixed stop at the ground board.

Airbrakes:

Backward: Adjustable stop at the horizontal pushrod, stops against the main bulkhead frame. Forward: Fixed stop, cross shaft lever stops at a tube piece.

6) Weights and Center of Gravity Positions:

After repairs, after installing of additional equipment, after new painting etc. there should be watched that the empty weight center of gravity is within the limits. If necessary balance weights are to be installed.

Empty weight		616	638	66o	682	7o5	lbs
center of gravity position	max.	21,45	21,o6	2o,68	2o,3	2o,o	
	min.	19,3	18,7	18,2	17,7	17,2	

behind datum point.

Leveling means: Tangate to rib 3 horizontal.

Datum: wing leading edge rib 3.

If the empty weight center of gravity is within the given limits it is verified that the in-flight center of gravity is correct provided the glider is properly loaded after the balance plate.

The center of gravity has an important effect to the gliders handling characteristics. Therefore one should pay attention to not exeed the given limits.

Too much backward position may become dangerous: Stalling and especially the spinning characteristics (flat spin) can be badly influenced. The elevator becomes more sensitive.

Too much forward position may deteriorate the performance and does not allow flying at maximum lift. (flare out when landing!).

The following ranges of flight position of. c.g. are tested:

a) max. forward position:
2,76 inches behind datum point.

b) max. aft position:
9,7 inches behind datum point.

6) Balancing instructions:

Cockpit load (Pilot + parachute).

single occup. front seat 143 - 220 lbs.

two occup. front seat 143 - 220 lbs.

Less load has to be completed with ballast on the seat (lead - or sand cushion).

Notice: If no parachute is carried a back cushion has to be used which has a compressed thickness of 4 inches.

Trim by weight

Fixing the 17 lbs standard trim weight at the forward foot board will compensate for 22 lbs pilot weight.

7) Approved Versions:

1) Rigging:

1) Clean and grease bolts and holes.
2) First put in left wing from the side and put in rear bolt. Do not tilt fuselage.
3) Put in right wing as mentioned under 2.
4) Put in the tapered pins (first the lower) and screw on the nut. By slight shaking the wing this can be done easier.
5) Secure the tapered pin with a safety needle. Secure the rear pins by the safety hooks.
6) Connect the aileron and airbrake linkages and secure by the safety needles.
7) Put on elevator, the front screw has to be fastened by the wrench.
There has to be watched that the bell crank is properly put into the pushrod counter - part. There is the danger of bending the push rod.
8) After the elevator has been put on the pushrod of the trim tab has to be connected to the tab bell crank by a pin, washer and safety pin. It is recommended to fix the push rod for transport.

2) Preflight Check

After rigging resp. daily befor the first flight:
Check all rigging connections for proper se - curing. Look for foreign bodies. Check the controls, airbrakes and the release hook for free movement.

It is recommended to check frequently the whole glider thoroughly. There have been found quite a lot of unsecured bolts and damage at such occasions. Use a flash light to look into every spot.

3) Derigging

Point 1 - 8 (Rigging) in reverse sequence. Grease all joints, danger of rust !

4) Road Transport

The design of a trailer is a special task and can not been talked here in all details. The firm of Schleicher will supply suitable drawings for trailers.
The wing is best supported at the spar root. The second support should have enough distance, best about 8 feet from wing tip. A sufficient wide aera at the supports is important.

The same considerations concerns to the fuselage. Fix points are the wheel, the fixing hole at the rear fuselage and the two welded in screw nuts on the side of the forward part of the fuselage.

Careful attention should be paid to prevent water coming into the interior of the wing and fuselage. The aileron and airbrake push rods in the fuselage should be fastened, best with a rubber strap.

5) Maintenance

Humity is the worst enemy of a wooden glider. Be careful to prevent water remaining in corners. On suspicion that water came into wings and fuselage, bring same into a dry room and turn every day. The glider is endangered especially on open trailers. In any case it must be taken care for that by covering the front part no splash water will touch the wing root.

Also by condensing water considerable quantities of humidity can enter the interior of the glider.

Strong sun irradiation will affect the finish by time, therefore the glider should not be exposed to the sun more than necessary. The treatment of the finish with good waxes and polishing material will increase the durability and improve the surface, an important fact for perfor - mance. The advantages of the laminarprofil can only be achieved by a smooth surface.

Sealing of gaps with adhesive tape will also cause some gain of performance. However at the cockpit caution is necessary, when parachute bail out shall be possible.

Cleaning of the plexiglas canopy only with suitable cleaners. If not available use pure water. Soft cloth (gloves cloth). In no case rub with hard cloth dry on plexiglass.

Lubrication of bearings:

So far as possible, the ball bearings are covered and therefore need no special mainte - nance. Only the bearings at the wing root, where the rigging connections do not allow a suitable protection, must be cleaned with gasoline when dirty and greased again.

The control surface bearings are to be dismantled and greased at the annual overhauls.

Wheel: Tube pressure 35 psi.

The c.g. hook especially is exposed to dirt and needs often cbaning and oiling.

The tail skid shoe has to be soled from time to time by welding on a new steel plate. But it should be removed for this work to prevent annealing the spring.

The pressure openings for the instruments at the fuselage are to be sealed with adhesive tape on transport or extended parking. During longer non-flying periods it is advisable to put the in - struments completely out and store it in a dry room. When installing again watch correct connecting of the tubes.

The safety belts are to be checked currently for fractures, damp-stain and corrosion.

6) Overhauls

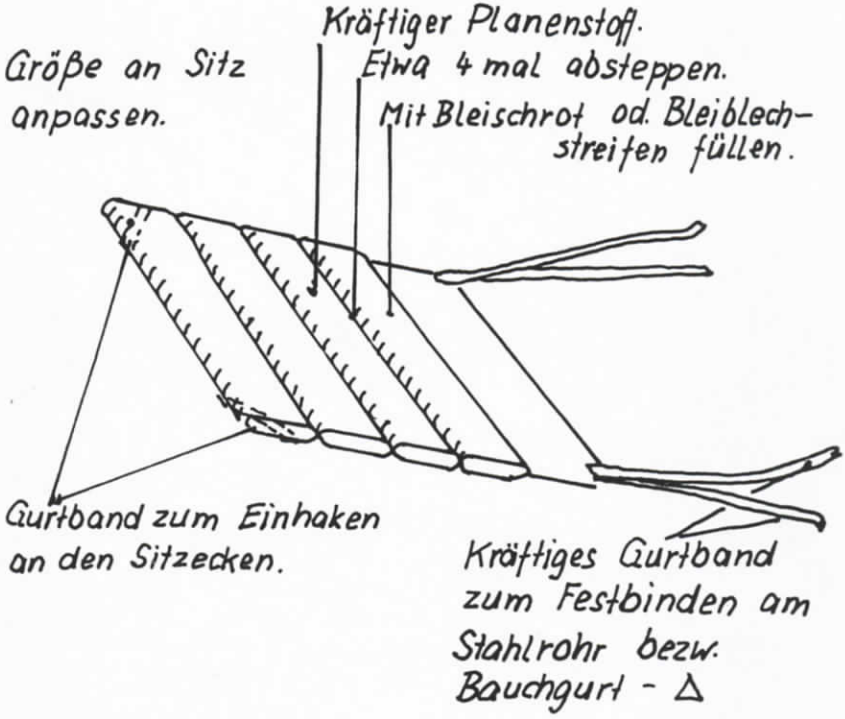
The tow hook has to be removed every 2ooo launches or 2 years and sent back to the manufacturer for overhaul.
The rudder cables are to be replaced if there are any signs of worth mentioning wear.

7) Repairs

All major repairs should be made in the manufacturing firm. In cases of doubt please ask the Schleicher firm.

8) Appendix

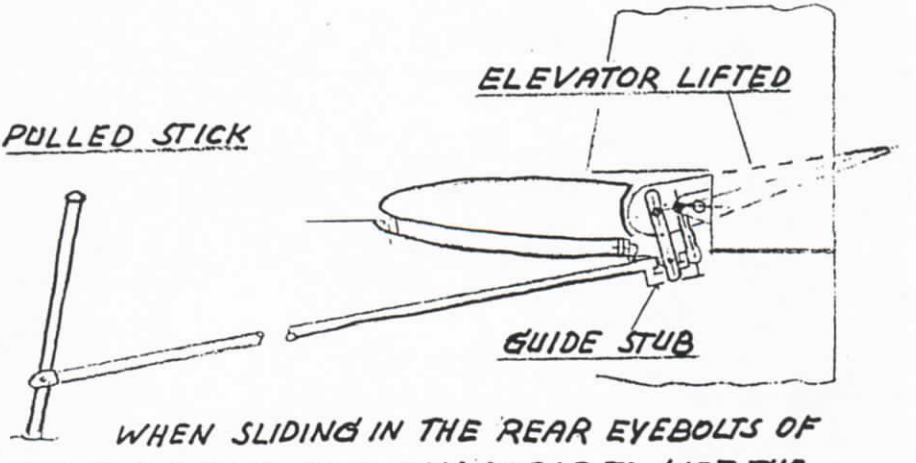
3 - View drawing
Balancing sheet.



2.11 Anlagen

1. Übersichtsblatt
2. Wägungsblatt
3. Schmierplan
4. Montageanweisung Höhenleitwerk

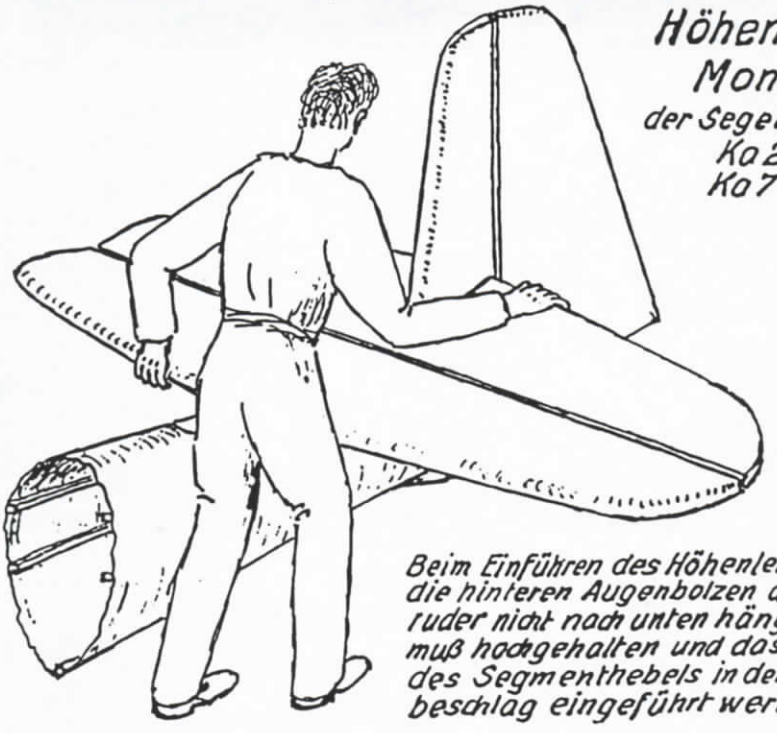
ELEVATOR UNIT
ASSEMBLY
SCHLEICHER K8



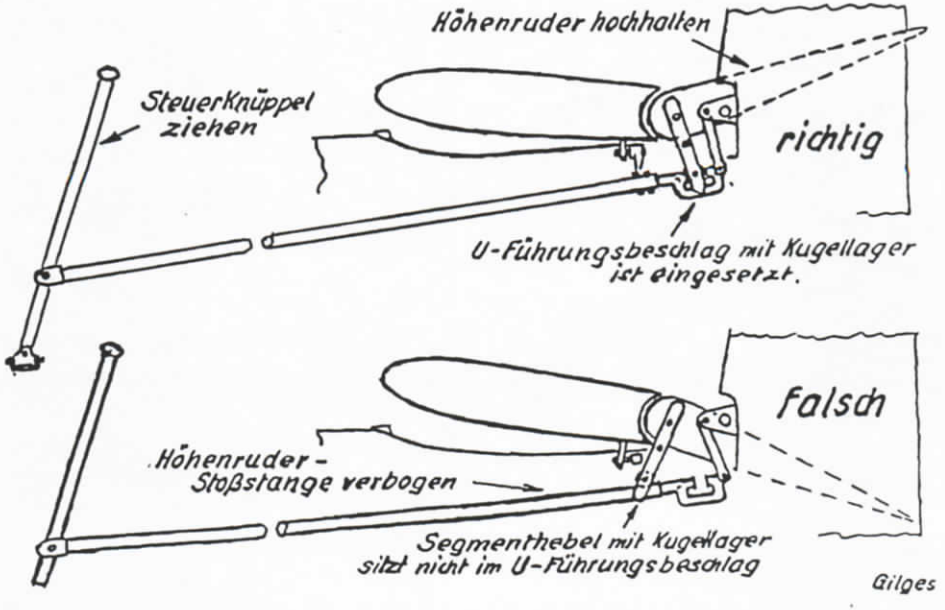
WHEN SLIDING IN THE REAR EYEBOLTS OF THE STABILIZER IT IS ADVISABLE TO LIFT THE ELEVATOR SOMEWHAT. THE BALL BEARING OF THE ELEVATOR CONTROL LEVER MUST FIT INTO THE GUIDE STUB OF THE PUSH-PULL TUBE TO AVOID THE RISK OF BENDING THE TUBE.

Höhenruder-Montage

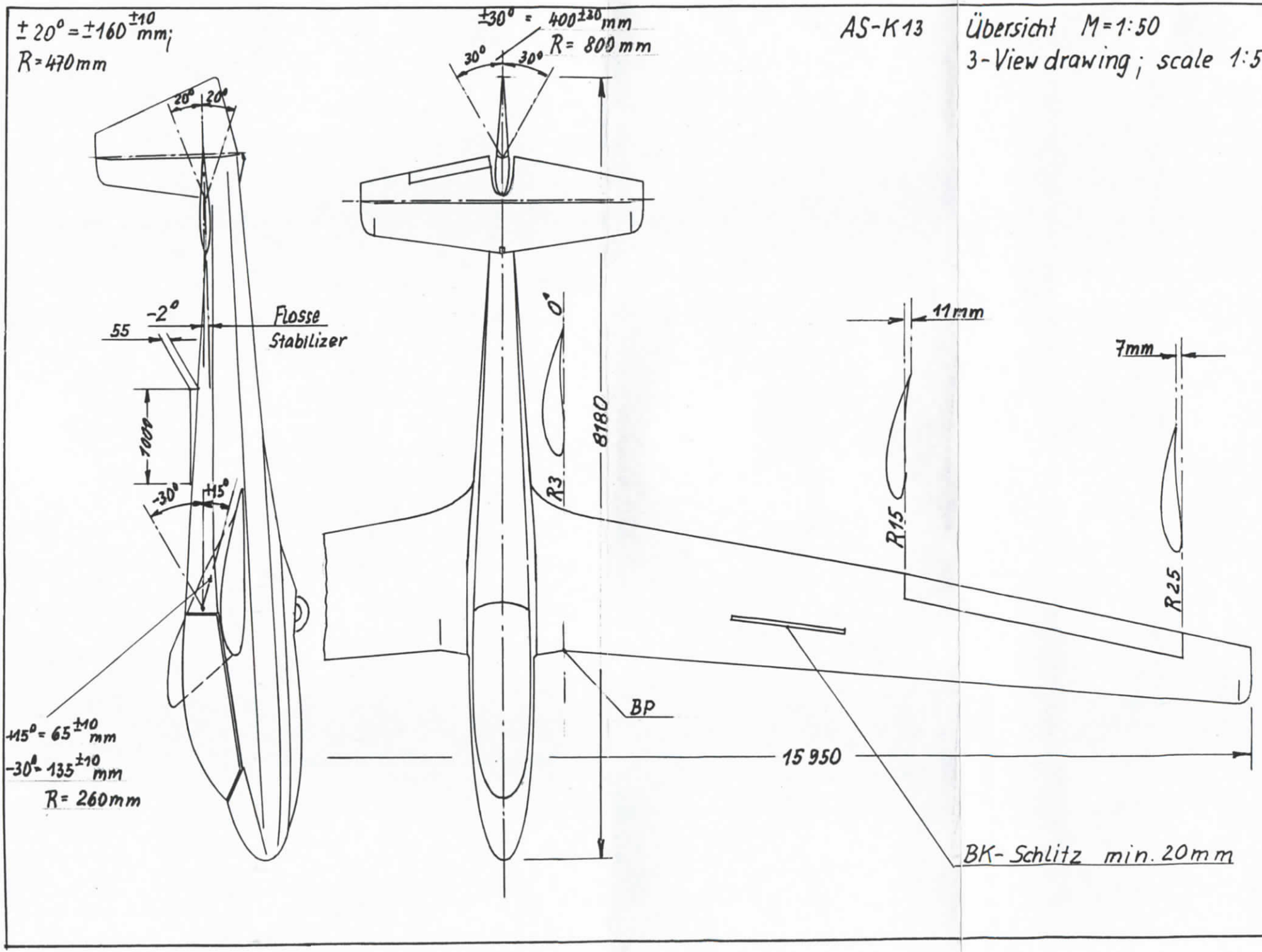
der Segelflugzeuge Ka 2b, Ka 6, Ka 7 und Ka 8.



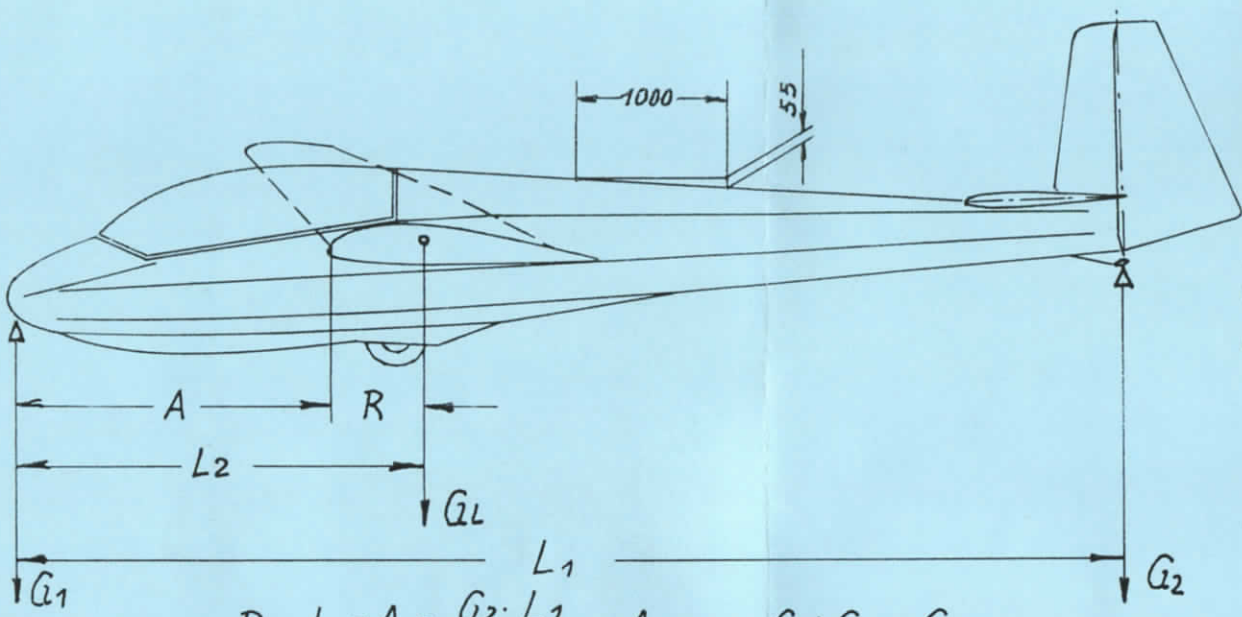
Beim Einführen des Höhenleitwerkes in die hinteren Augenbolzen darf das Höhenruder nicht nach unten hängen, sondern muß hochgehalten und das Kugellager des Segmenthebels in den U-Führungsbeschlag eingeführt werden!



Gilges


± 20° = ± 160 ±10 mm;
R = 470 mm
±30° = 400 ±10 mm
R = 800 mm
AS-K13
Übersicht M = 1:50
3-View drawing; scale 1:50
20°
20°
30°
30°
-2°
Flosse
Stabilizer
55
100
-30°
+15°
+15° = 65 ±10 mm
-30° = 135 ±10 mm
R = 260 mm
0°
R3
8180
11mm
7mm
R15
R25
BP
15 950
BK-Schlitz min. 20mm

AS-K13 Wägeblatt; Balancing sheet



$$R = L_2 - A = \frac{G_2 \cdot L_1}{G_L} - A \qquad G_1 + G_2 = G_L$$

G_L = Leergewicht; empty weight.
G_1 = Gewicht am vorderen Auflagepunkt
weight at the front support.
G_2 = Sporngewicht; weight at tail skid.
R = Schwerpunkt rücklage.
center of gravity position.

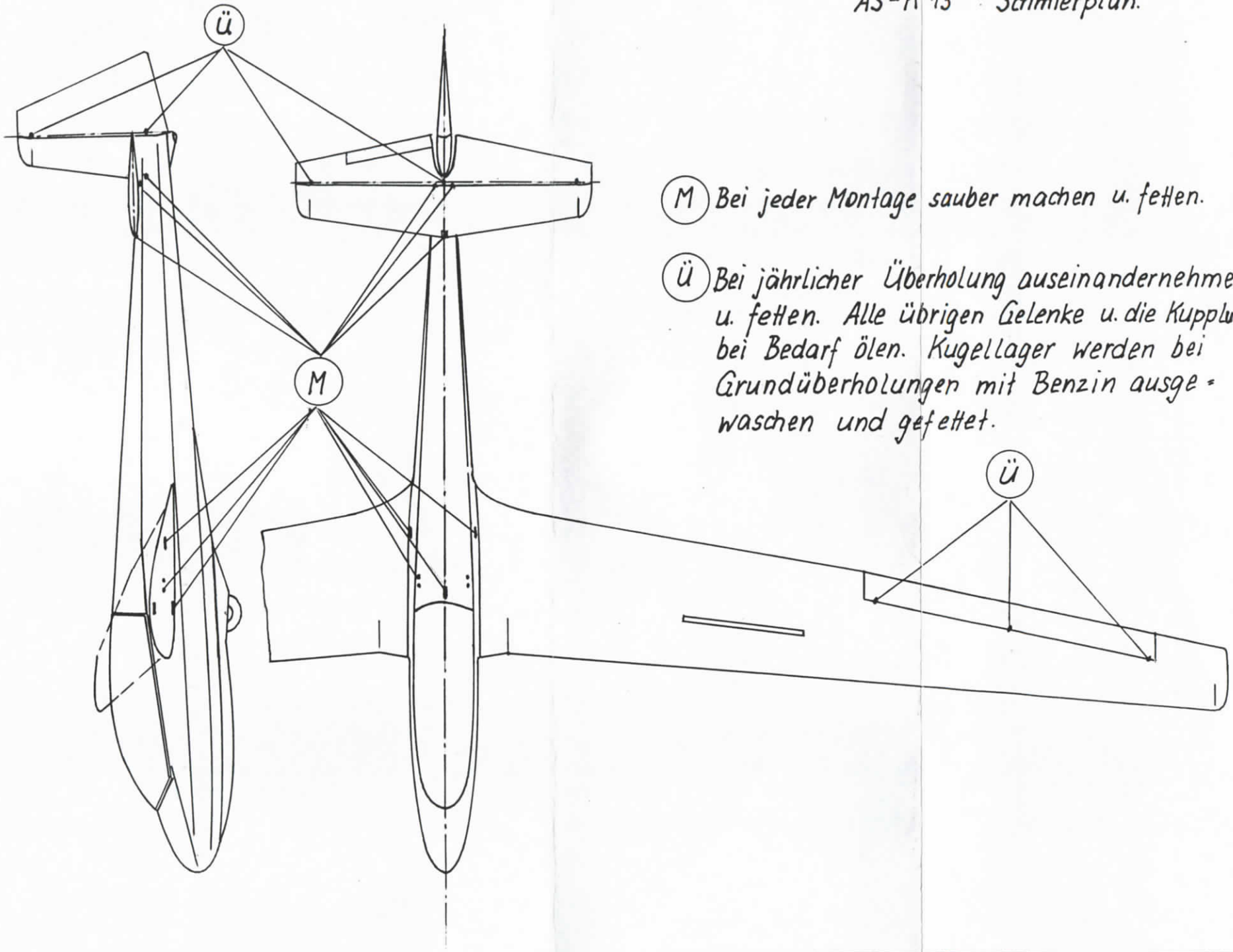
G_L = Leergewicht; empty weight	280	290	300	310	320	kp	330*	340*
R { Vorderste Schwerpunktlage; most forward position of C.of.Gr. empty min.	490	475	462	449	437	mm	426	415
R { Hinterste Schwerpunktlage; most rearward position of C.of.Gr. empty max.	545	535	525	516	508	mm	499	492

erweitert
18.05.92
GW/Jus

* Flugzeug möglichst erleichtern, mögliche Gesamt-Zuladung sonst zu gering für einen Doppelsitzer!

Lighten the sailplane as much as possible. Otherwise max. payload too low for a double seater!

AS-K 13 Schmierplan.



(M) Bei jeder Montage sauber machen u. fetten.

(Ü) Bei jährlicher Überholung auseinandernehmen u. fetten. Alle übrigen Gelenke u. die Kupplung bei Bedarf ölen. Kugellager werden bei Grundüberholungen mit Benzin ausgewaschen und gefettet.

Semiaerobatics:

Besides spins the following aerobatic maneuvres may be executed:

Loops

Stall turns

Lazy eights

Loop: Speed of entry 90 to 100 mph.

Stall turn: Speed of entry 100 to 110 mph. During climb at approx. 60 mph there has to be applied full rudder and slightly opposite aileron.

Lazy eight: Speed at the crossing point 90 to 100 mph.

Blatt 1 von 2	Technische Mitteilung für Technical Note for	**Alexander Schleicher** GmbH & Co. Segelflugzeugbau D - 36163 Poppenhausen

Muster
Type:

Ka 6 ..... TM-Nr. 25
K 7 ..... TM-Nr. 22
K 8 ..... TM-Nr. 28
ASK 13 ..... TM-Nr. 18
ASK 16 ..... TM-Nr. 15
ASK 18 ..... TM-Nr. 8

Gegenstand:
Subject

Inspektion des mittleren Höhenruderlagers
Inspection of the central support of the elevator

Betroffen:
Applicability:

Ka 6 ..... Kennblatt *Type Certificate* LBA 205,
Baureihen *Variants* Ka 6, Ka 6/0, Ka 6B, Ka 6BR, Ka 6C, Ka 6CR, Ka 6B-S
K 7 ..... Kennblatt *Type Certificate* LBA 211
K 8 ..... Kennblatt *Type Certificate* LBA 216,
Baureihen *Variants* K 8, K 8B, K 8C
ASK 13 ..... Kennblatt *Type Certificate* LBA 267
ASK 16 ..... Kennblatt *Type Certificate* LBA 758
ASK 18 ..... Kennblatt *Type Certificate* LBA 307,
Baureihen *Variants* ASK 18, ASK 18B

Bei Ka 6: alle Werk-Nummern mit gedämpften Leitwerk (Pendelleitwerk nicht betroffen).
Bei den übrigen Typen: alle Werk-Nummern.
All Ka 6 with fixed horizontal stabilizer (no all moving tails affected).
Of the other types: All serial numbers

Klassifizierung:
Classification:

Geringfügige Änderung
Minor Change

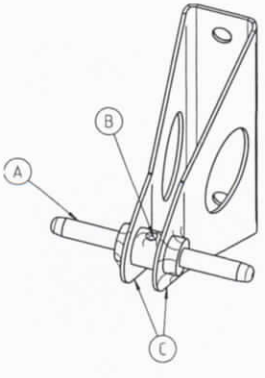
Dringlichkeit:
Compliance:

Vor der nächsten Jahresnachprüfung
Before the next annual inspection

Grund:
Reason:

Bei einem Segelflugzeug wurde festgestellt, daß die Achse (A) des Höhenruderlagers lose war. (siehe Fig. 1)

Fig. 1



Die Achse ist in einer Buchse geführt. Durch ein Loch (B) in dieser Buchse ist sie festgeschweißt. Diese Verbindung hat nicht mehr bestanden.
On one glider it was found, that the axis (A) of the central support of the elevator was loose. (see fig. 1). This axis is supported in a bushing. Through a hole (B) in this bushing, the axis is fastened by a weld. This welded connection did not exist any more.

Technische Mitteilung für
Technical Note for

Alexander Schleicher
GmbH & Co.
Segelflugzeugbau
D - 36163 Poppenhausen

Maßnahmen:
Action:

Das Höhenruder demontieren. Prüfen, ob die Lagerachse (A) fest sitzt.
Dabei sind zusätzlich auch die Bleche des Lagerbocks rund um die Schweißnähte (C) auf Risse zu prüfen (mindestens mit einer Lupe 10facher Vergrößerung).
Gegebenenfalls ist der Lagerbock zu ersetzen.
Diese TM ist als Anhang in das Flug- und Betriebshandbuch einzufügen und im Berichtigungsstand einzutragen.
Unmount the elevator. Inspect the axis (A). It must have a firm seat.
At that occasion, the sheet metal of the support is to be checked for cracks around the welds (C) (at least with a 10x magnification loupe).
When indicated, the support must be replaced.
This copy of the Technical Note must be inserted in the Flight and Operations Manual of the respective glider as an annex, and a corresponding entry must be made into the "Amendments to the Manual"

Material und Zeichnungen:
Material and Drawings:

Falls Ersatz bestellt werden muß, lauten die Bestellnummern:

Ka 6	L-205.33-U170, Mittleres Höhenruder-Lager
K 7	L-211.33-U08, Mittleres Höhenruder-Lager
K 8	L-216.32-U08, Mittleres Höhenruder-Lager
ASK 13	130.33.0130, Mittleres Höhenruder-Lager
ASK 16	160.33.0005, Mittleres Höhenruder-Lager
ASK 18	L-307-33.0003, Mittleres Höhenruder-Lager

In case replacement must be ordered, the part.-numbers are listed above

Masse und Schwerpunktlage:
Weight (Mass) and Balance:

Kein Einfluß
not effected

Hinweise:
Notes:

Es sei dabei noch an die LTA 72-7/3 erinnert, die in dreijährlichen Intervallen die Überprüfung der Verleimung am Höhenruder verlangt (außer ASK 16). Wir empfehlen diese Maßnahme bei jeder Durchführung der LTA 72/3 zu wiederholen.
Alle Maßnahmen sind von freigabeberechtigtem Personal entsprechend EU-VO. 2042/2003 Teil M / Teil 66 [1] zu prüfen und in den Prüfunterlagen sowie im Bordbuch eine Freigabe zu bescheinigen. Die Ergänzung des Handbuchs ist im Berichtigungsstand und im Verzeichnis der Handbuchseiten einzutragen, dies kann der Halter durchführen.

At this opportunity, we remind of AD LTA 72-7/3, which requires the inspection of the glue joints at the elevator in 3-year intervals (with the exception of the ASK 16). We recommend to repeat this action every time the actions of AD LTA 72-7/3 are carried out.
All actions are to be inspected by certifying staff according to Commission regulation (EC) 2042/2003 Part M / Part 66 [2], and have to be certified in the sailplane inspection documents and in the sailplane logbook. The amendment of the manual must be entered on the page „Record of Revisions" and in the „List of effective pages", this can be done by the operator.

Poppenhausen, den 25.05.2010

Alexander Schleicher
GmbH & Co.

i.A. M. Greiner
(Michael Greiner)

Diese Änderung wurde mit Datum vom 17.05.10 durch die EASA mit der Änderungsnummer 10030052 anerkannt.
This Change has been approved by the EASA at the date of the 17.05.10 with the Minor Change Approval 10030052.

[1] Solange keine Festlegungen für freigabeberechtigtes Personal für Segelflugzeuge und Motorsegler getroffen wurden, gelten noch die einschlägigen Vorschriften des Mitgliedstaates (§66.A.100).
[2] *As long as no provisions for certifying staff for sailplanes and powered sailplanes were laid down, relevant legislation of the member states is applicable (§66.A.100).*

SHEET: 1 of 3	Technical Note for	Alexander Schleicher GmbH & Co. Segelflugzeugbau D-6416 Poppenhausen

Glider model:

Ka 2 u. Ka 2B	TN-No. 11
Ka 6, 6/0, 6B, 6BR, 6CR, 6B-S	TN-No. 21
K7	TN-No. 18
K8, K 8B, K 8C	TN-No. 23
K9	TN-No. 1
K11	TN-No. 1
ASK 13	TN-No. 12
ASK 18, ASK 18B	TN-No. 6

Serial number applicability:

Ka 2,	Data-Sheet No. 140,	all serial no.s
Ka 2B,	Data-Sheet No. 203,	all serial no.s
Ka 6,	Data-Sheet No. 205,	all serial no.s
Ka 6/0,	Data-Sheet No. 205,	all serial no.s
Ka 6B,	Data-Sheet No. 205,	all serial no.s
Ka 6BR,	Data-Sheet No. 205,	all serial no.s
Ka 6CR,	Data-Sheet No. 205,	all serial no.s
Ka 6BS,	Data-Sheet No. 205a,	serial no. E1
K7,	Data-Sheet No. 211,	all serial no.s
K8,	Data-Sheet No. 216,	all serial no.s
K8B,	Data-Sheet No. 216,	all serial no.s
K8C,	Data-Sheet No. 216,	all serial no.s
K9,	Data-Sheet No. 221,	serial no. 1
K11,	Data-Sheet No. 668,	serial no. V1
ASK 13,	Data-Sheet No. 267,	all serial no.s
ASK 18,	Data-Sheet No. 307,	all serial no.s
ASK 18B,	Data-Sheet No. 307,	all serial no.s

Subject:

Elevator.

Compliance:

Prior to the next take-off.

Reason:

A glider of the model K7 failed to gain normal flight attitude immediately after tow rope release on winch launch. With the stick full back only the left elevator could be actuated in the correct direction; the right elevator deflected downwards. The reason for this was a loose glue bond at the elevator rib 1 at which the elevator fitting is attached. Similar incidents lead already before to the issue of the LTA 72-7 dated Feb.9,1972.

Action:

1. Remove elevator.
   Check that the glued joint between rib 1 and the leading edge plywood and the elevator spar respectively is in good condition (see Fig.1). Before doing so check whether the LTA 72-7 of Feb.9, 1972 was already previously accomplished (this is not applicable to K9, K11 and ASK 18); if yes then the fabric strip first carefully has to be detached in order to be able to check the glued joint.

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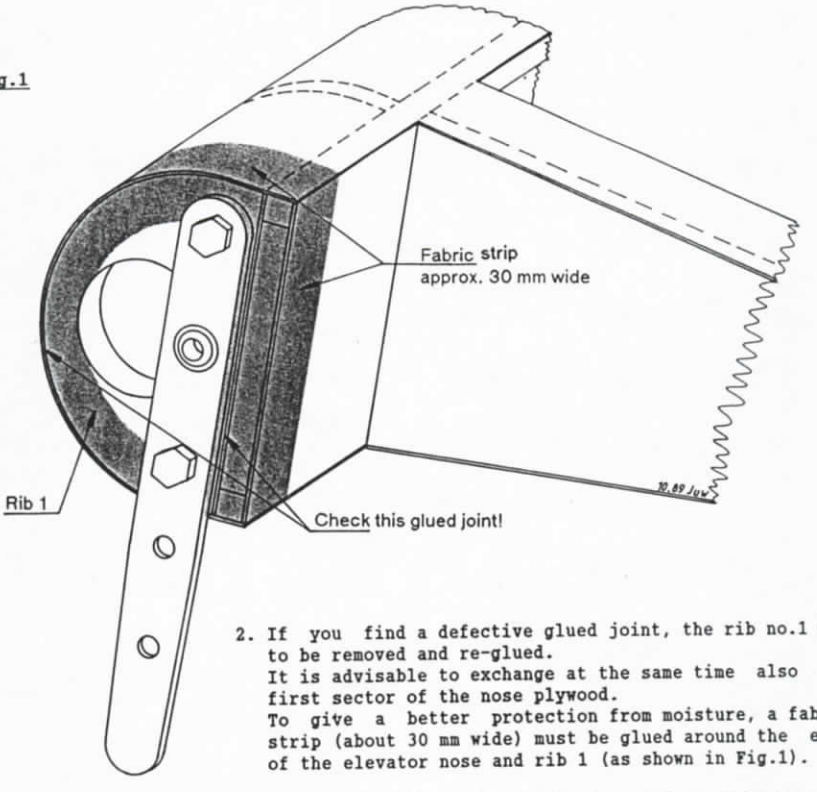
Technical Note
for
Glider Models as per Sheet 1

Alexander Schleicher
GmbH & Co.
Segelflugzeugbau
D-6416 Poppenhausen

Fig.1



2. If you find a defective glued joint, the rib no.1 has to be removed and re-glued.
It is advisable to exchange at the same time also the first sector of the nose plywood.
To give a better protection from moisture, a fabric strip (about 30 mm wide) must be glued around the edge of the elevator nose and rib 1 (as shown in Fig.1).

3. The above action under points 1. and 2. must be repeated every three years during the annual re-inspection. This copy of the Technical Note must be inserted in the Flight and Operations Manual of the respective glider as an annex and a corresponding entry must be made into the "Amendments to the Manual".

Material & drawings:

Rib 1 made from multi-plywood, 15 m thick, and nose plywood, 1 mm thick, according to DIN L 182/183, class 1/2 or NL 9128, 6.1013. Drawing as above.

Mass and C.G.:

It is not necessary to redetermine the mass and C.G. data.

Technical Note
for
Glider Models as per Sheet 1

Alexander Schleicher
GmbH & Co.
Segelflugzeugbau
D-6416 Poppenhausen

Notes:

Actions 1. and 3. can be accomplished by a person who is familiar with such work.
Action 2. must only be accomplished by a technical aviation service station holding an appropriate license; the accomplishment of all actions must be certified by a licensed aviation inspector in the glider logbook and in the inspection certificates.

Poppenhausen, October 4, 1989

ALEXANDER SCHLEICHER
GmbH & Co.

i.A. L.-W. Jumtow

L.-W. Jumtow.

The German original of this Technical Note has been approved by the LBA under the date of Oct.17, 1989 (signature: FRIESS). The translation into English has been done by best knowledge and judgement; in any case of doubt the German original is controlling.

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AIRWORTHINESS DIRECTIVE

72-7/3 Schleicher

Date of issue:

13. Dez. 1989

Affected Sailplane:
German Type Certificate

No. 140,	Ka 2,	all serial nos.
203,	Ka 2B,	all serial nos.
205,	Ka 6,	all serial nos.
205,	Ka 6/0,	all serial nos.
205,	Ka 6B,	all serial nos.
205,	Ka 6BR,	all serial nos.
205,	Ka 6CR,	all serial nos.
205a,	Ka 6BS,	serial no. E1
211,	K7,	all serial nos.
216,	K8,	all serial nos.
216,	K8B,	all serial nos.
216,	K8C,	all serial nos.
221,	K9,	serial nos. 1
668,	K11,	serial No. V1
267,	ASK 13,	all serial nos.
307,	ASK 18,	all serial nos.
307,	ASK 18B,	all serial nos.

Subject:
Elevator

Reason:
Loose glue joints on rip 1 of the elevator

Action:
In accordance with the respektive Technical Note

Compliance:
Before the next start

Technical publications of the manufacturer:
Alexander Schleicher, Technical Note, October 4, 1989 "Elevator"

Model	Ka 2 and Ka 2B	TN No. 11
	Ka 6, 6/0, 6B, 6BR, 6CR, 6B-S . .	TN No. 21
	K7	TN No. 18
	K8, K 8B, K 8C	TN No. 23
	K9	TN No. 1
	K11	TN No. 1
	ASK 13	TN No. 12
	ASK 18, ASK 18B	TN No. 6

which become herewith part of this AD and may be obtained from Messrs.
Alexander Schleicher GmbH & Co. Segelflugzeugbau,
D-6416 Poppenhausen, Wasserkuppe, Federal Republic of Germany

Accomplishment and log book entry:
Action 1 and 3 to be accomplished by a skilled person.
Action 2 to be accomplished by an approved service station.
The accomplishment of this AD must be certified by a licensed inspector in the powered gliders inspection documents and in the log-book.

Note:
This Airworthiness Directive replaces AD-No. 72-7/2 of August 24, 1989.

SHEET: 1 of 2	ASK 13 Technical Note No. 14	Alexander Schleicher GmbH & Co. Segelflugzeugbau D-6416 Poppenhausen

Subject: Inspecting the bearing brackets and the toggle force of the airbrake control circuit.

Serial number applicability: Glider ASK 13, Data Sheet no.267.
A) Serial no.s 13000 thru 13689 including;
B) All serial no.s.

Compliance:
A) The action must accomplished with the next annual C. of A. inspection, but before or on March 31, 1992, at the latest.
B) The action must become part of each future annual C. of A. inspection.

Reason: On some gliders of the model ASK 13 the bearing brackets of the airbrake control circuit have broken at the wing root rib. The failure was caused by too high toggle forces which resulted from a wrong adjustment of the toggle in the airbrake control circuit.

Action:

1. Carefully inspect the bearing brackets of the airbrake control circuit at the wing root ribs as well as the root ribs themselves for damage, cracks etc. In case of doubt the paint must be removed off the bracket. In most cases the breaks were found at the bottom of the front bearing bracket; refer to the drawing on Sheet 2 of this TN !
   Where damages at one bearing bracket are found, this part must be exchanged; where a root rib is damaged, it requires repair.

2. Check the toggle force as described under Point 3.) and where necessary adjust.

3. Adjustment Instructions For Airbrake Toggle

3.1 Undo the airbrake return spring at the front operating lever in the cockpit.

3.2 The airbrake toggle - each side separately - must be adjusted such that a force of approx. 3 to 4 daN is obtained at the front airbrake operating lever when measured from above (measurement distance from the pivot point of the operating lever towards the hand grip = 300 mm).

3.3 Adjustment of the toggle force is done by turning in or out the adjusting heads in the fuselage. If the correct adjustment values cannot be achieved here, the thickness of the airbrake stop blocks inside the wings must either be increased or cut back.

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3.4 Take care that the toggle must not go over the dead point which would cause the airbrakes to extend again.

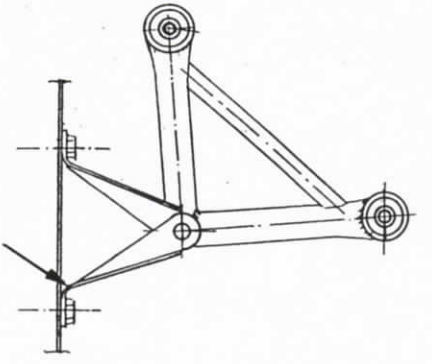
3.5 For both airbrakes together, the hand forces must not exceed approx. 6 daN, when measured at the front airbrake operating lever from above.

3.6 Re-fit the airbrake return spring at the front operating lever in the cockpit.

4. This Technical Note must be inserted as appendix into the ASK 13 Flight and Maintenance Manual and this action must be entered into the "Amendments" on page 2.

5. The inspection of the bearing brackets and of the toggle force of the airbrake control circuit (as described under points 1. thru 3.) must be included from now on within each annual C. of A. inspection.

Material & drawings:



Notes:

The actions under Point 1.) thru 3.) can be accomplished by any competent person. The accomplishment of this mod must be certified by a licensed aviation inspector in the glider`s inspection documents and in the log-book.
The action under Point 4.) can be accomplished by the owner / operator of the glider himself.

Poppenhausen, September 27, 1991

ALEXANDER SCHLEICHER
GmbH & Co.
i.A. Lutz-Werner Jumtow.

The German original of this Technical Note has been approved by the LBA under the date of October 10, 1991 (signature: SKOV). The translation into English has been done by best knowledge and judgement; in any case of doubt the German original is controlling.

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